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AN INVESTIGATION OF HIGH-TEMPERATURE VACUUM AND
HYDROGEN FURNACE BRAZING

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SUMMARY

The vacuum and the hydrogen brazing of four heat-resistant alloys with two types of high-temperature brazing alloy were investigated. The effect of time at two brazing temperatures on the 1200° F shear strength of joints and on the base-metal properties was studied.

Brazing techniques were evaluated for alloys that can be age hardened and that contain titanium and aluminum in a vacuum as well as in dry hydrogen.

In general, results showed that of the two brazing alloys used, the boron-free alloy was less damaging to base metal than the boron-bearing alloy, but that shear joints made with the boron-bearing braze were stronger. Although it was thought that the primary difference between the alloys was boron content, the higher carbon of the boron-bearing alloy may be significant.

Furnace brazing temperatures and time at temperatures were important factors in lowering the tensile strength and elongation of braze-coated sheet-metal tensile specimens. The effects varied depending on the base metal and the brazing alloy used.

Shear specimens of all four base alloys brazed in hydrogen with both types of brazing alloy exhibited erratic joint coverage by the brazing alloys. The data indicated, however, that if joint coverage was complete, vacuum and hydrogen brazing produced joint shear strengths of about the same magnitude.

INTRODUCTION

During the last five years, interest in high-temperature brazing of heat-resistant alloys has been growing steadily. Advancements in dry-hydrogen furnace-brazing techniques have attracted the attention of engineers and designers in many fields of engineering. With furnace brazing it is possible to fabricate intricate parts and assemblies that would be otherwise impossible. The use of light-weight sheet-metal components for turbojet engines, now being evaluated by research groups throughout the country, depends largely on successful furnace brazing.

High-temperature furnace brazing of thin sheet-metal assemblies at the NACA Lewis laboratory revealed a serious problem. Though little difficulty was encountered in obtaining adequate joint coverage and joint shear strength, testing and subsequent metallographic studies revealed damaging effects on the base metal. The diffusion of the braze alloy into the base metal resulted in a considerable reduction in the strength and ductility of the base metal. This undesirable effect on base-metal strength and ductility indicated a need for changes in the furnace brazing process, or, perhaps a less susceptible base metal or a less damaging braze composition.

Consequently, a correlative study was designed with the following purposes:

- (1) To investigate the effect of various brazing temperatures on the tensile strength of sheet specimens coated with braze and on the shear strength of brazed joints
- (2) To investigate the effect of time at the brazing temperature on the tensile strength of sheet specimens coated with braze and on the shear strength of brazed joints
- (3) To determine ways to minimize base-metal damage and yet provide adequate shear joint coverage and strength

Four base metals, two commercial brazing alloys, two brazing temperatures, and two controlled times at each brazing temperature were investigated. These variables were studied in both vacuum and hydrogen brazing atmospheres.

APPARATUS AND MATERIALS

Brazing Alloys

Silicon and boron are additives for reducing the melting temperature of heat-resistant brazing alloys. The boron-bearing brazing alloys available are essentially nickel-chromium alloys with various percentages of silicon and boron added. Nicrobraz (AMS-4775) was used throughout this study to represent the boron-bearing type of brazing alloy. Alloy, G.E. 81, was used throughout to represent the boron-free type of brazing alloy and contains only silicon so that the liquidus temperature of the nickel-chromium mixture was lowered. It should be noted that Nicrobraz also has a higher carbon content than G.E. 81, and some of the effects described herein may result from this difference. Compositions of both brazing alloys are given on table I.

Nickel-chromium-type brazing alloys possess good oxidation resistance up to 1600° F and retain adequate joint shear strengths up to a test

temperature of 1200° F. A shear strength of 30,000 pounds per square inch at 1200° F for S-590 alloy specimens brazed with Nicrobraz is reported in reference 1. The Nicrobrazing in reference 1 was done at a temperature of 2150° F for 30 minutes in dry hydrogen.

Base Metals

The following base metals were used in this investigation: A-286 (AMS-5525), Inconel X (AMS-5542), N-155 (AMS-5532b), and L-605 (AMS-5537).

The compositions of these base metals are given in table I.

One of the first questions asked by designers is whether or not base-metal damage varies with the type of base metal. Since this was believed quite likely, four heat-resistant alloys were selected for investigation. All four base metals chosen for this program are used frequently in high-temperature sheet-stock application.

Two of the base metals selected, N-155 (AMS-5532b) and L-605 (AMS-5537), can be brazed without difficulty in either a dry-hydrogen atmosphere or vacuum.

Materials that can be age hardened such as A-286 (AMS-5525) and Inconel X (AMS-5542), have received considerable attention lately for use in air-cooled turbine designs. Unfortunately, the alloys containing aluminum and titanium are difficult to braze. Both the dry-hydrogen atmosphere at a dewpoint of -60° to 80° F and a vacuum with a pressure of 1 to 4 microns of Mercury were inadequate to braze these alloys. The oxides formed on the surface during the brazing cycle prevent flow and wetting by the brazing alloy.

The stability of the undesirable oxides on the base-metal surfaces at the brazing atmosphere conditions and temperature is the deciding factor in the wetting of the base metal by the brazing alloy. Chromium oxide, for example, may be stable and actually form at a -80° F dewpoint in a hydrogen atmosphere at 1000° F. However, as the temperature rises the equilibrium changes, and the oxides are reduced before the brazing temperature is reached. The reduction temperature for chromium oxide is lowered if the partial pressure of the oxygen is reduced. But up to the time of this investigation, it was impracticable to lower the dewpoint (hydrogen brazing) or pressure (vacuum brazing) enough to reduce the oxides on aluminum- and titanium-bearing alloys. One dependable way to braze these base metals is to plate them with a metal whose oxide can be reduced at temperatures lower than the brazing temperature being used.

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Plating of Base Metals

Iron and nickel plate were found satisfactory in brazing base metals. Nickel plating was used throughout this study, because it was more readily available at commercial platers. The "electroless" type of nickel plate (ref. 2) as well as the electrolytic type were used in this evaluation. As pointed out in reference 2, electroless plate can be applied uniformly to complex as well as internal surfaces. The advantage of electroless plating intricate shapes and internal surfaces with a uniform thickness would be of considerable value. Preliminary tests in a vacuum showed that the electroless plate used in this investigation melted at approximately 1700° F. This temperature is well below the brazing temperature, but the plate still retained its protective value. Electrolytic nickel plate is essentially pure nickel (m.p., 2642° F), whereas electroless nickel contains up to 10 percent phosphorous. The percent of phosphorous varies with the plating conditions.

SPECIMENS

Shear Specimens

It is considered very important that if comparisons are made between different brazing conditions or different brazing alloys, that they are made from results obtained from the same kind of specimen.

Probably the most important single factor affecting the shear joint strength of a brazing alloy is the degree of joint coverage. The variables, brazing temperature, time at temperature, brazing alloy and base-metal compositions, and atmospheres used in this investigation can all affect the degree of joint coverage. The shear specimen used (figs. 1 and 2) was designed with a relatively large shear area so that the effect of poor joint coverage would be increased. The specimen was so designed that the tensile yield strength of any base metal would not be exceeded during testing by an expected maximum shear joint strength of 40,000 pounds per square inch at 1200° F.

Joint clearance and the amount of brazing alloy present affect joint coverage as well as the actual joint shear strength (ref. 3). Machining tolerances were specified to hold joint clearances between 0.0015 and 0.0025 inch. Clearances were altered on specimens to be plated to ensure the final desired clearances. Samples to be tested were brazed using different amounts of brazing alloy powder placed in the bore of the specimen. A 0.40-gram sample of brazing alloy powder was used on the shear specimens although it provided a slight excess. With the brazing alloy so placed within the specimen, it was possible to heat the specimen to the brazing temperature ahead of the brazing alloy. To heat the specimen ahead of the brazing alloy is considered better than the brazing alloy temperature

exceeding the specimen temperature during heating. Heating the specimen to the brazing temperature first, will minimize the time the base metal must be exposed to molten braze alloy directly under the applied braze. Whenever feasible, the brazing alloy should be applied in an area of low stress and allowed to flow to areas of high stress.

To measure the shear strength, all excess braze and fillets were removed. The removal of excess braze and fillets from the specimen used in this program involved simple boring and necking machining operations (see fig. 1).

Tensile Specimens

The sheet-metal specimen shown in figure 3 was used to evaluate the effect of brazing on base-metal strength and ductility. The brazing alloy powder could not be applied dry to sheet-metal specimens. Therefore, a heavy slurry of braze powder and acryloid (thinned with acetone) was applied to one side of the specimen. When dry, the braze was trimmed in order to leave a layer 0.015 inch thick and 1/4 inch wide across the center of the 1-inch test section. Three specimens of each base alloy with no brazing alloy and three of each base alloy with each brazing alloy were prepared.

Furnaces

Vacuum furnace. - The furnace used for all the vacuum brazing reported in this investigation is shown in figure 4. The vacuum furnace employs a 100-kilowatt motor-generator induction heater to heat a 9-inch diameter graphite crucible (susceptor), which, in turn, heats the work load within it by radiation.

The heating coil assembly of the furnace was designed so that radiation losses would be high, assuring a rapid drop in temperature at the end of the heating cycle.

A 500-liter-per-second oil diffusion pump and a 100 cubic feet per minute mechanical pump maintains a pressure of 1 to 4 microns of mercury throughout the brazing cycles. A Pirani-type gage is used to measure vacuum.

Hydrogen furnaces. - The hydrogen brazing was done commercially with a batch-type furnace and a continuous hump-type furnace. Atmospheres reported for all cycles had a dewpoint of -70° to -80° F.

PROCEDURE

Brazing

In this study the effect of various times at brazing temperatures of 2100° and 2150° F on the shear strengths of brazed joints and on the degree of base-metal damage was investigated. The selection of these two temperatures (2100° and 2150° F) was based on past experiment at the Lewis laboratory with the two brazing alloys investigated (Nicrobraz and G.E. 81). It was found that incomplete joint coverage for large joint areas was likely to occur if brazing was done below 2100° F and that excessive parent-metal damage in thin sections could be expected above 2150° F. Two brazing atmospheres were investigated, vacuum and hydrogen atmosphere, the details of which will be covered in subsequent sections. A summary of all the brazing variables investigated is shown in table II.

Vacuum brazing. - A typical brazing cycle consisted of (1) evacuation of the furnace by the vacuum pumps described earlier to a pressure of approximately 1 micron of mercury, (2) slow heating of the work load at a rate of approximately 50° F per minute to 1850° F, (3) holding at 1850° F for 10 minutes, (4) rapid heating (150° F/min) to the selected brazing temperature, (5) holding the brazing temperature for the selected time, (6) cutting of the electric power, which results in a very rapid cooling (300°/min) to 1800° F, and (7) a diminishing rate of cooling to room temperature.

To ensure uniform temperatures, the shear specimens were placed in a single concentric circle within the furnace susceptor (fig. 5). The thermocouple was placed within one of the specimens as close as possible to the brazing alloy. All brazing temperatures were controlled to $\pm 5^\circ$ F. At the brazing temperature, the measured pressure ranged from 3 to 4 microns of mercury. A pressure of 3 to 4 microns of mercury was found in preliminary tests to be adequate for brazing N-155 and L-605 alloys. Also, it was found that wetting of joints and flow into joints were possible for unplated Inconel X but not for unplated A-286. These same tests in a -80° F dewpoint hydrogen atmosphere revealed very poor wetting on both Inconel X and A-286. On the basis of these tests, it was decided to evaluate both plated and unplated Inconel X specimens for vacuum brazing and only plated specimens for hydrogen brazing. The electroless type of plate was chosen for Inconel X, whereas A-286 specimens were evaluated with electroless as well as electrolytic nickel plate.

In order to ascertain accurately the effect of time at the brazing temperature used in the brazing cycle, it was considered very important to have sufficient electric power available to raise the temperature of the work load from 1850° F to the desired brazing temperature as rapidly as possible. Slow heating through the melting range of the brazing alloy means a longer exposure of the base metal to the molten braze and can be

expected to have more damaging effects on the base metal than rapid heating. Flow and joint coverage (ref. 4) were adversely affected by slow heating through the melting range of a given brazing alloy. Heating slowly to 1850° F, holding for 10 minutes, and then heating more rapidly to the brazing temperature proved satisfactory for sheet-metal assemblies in that no appreciable distortion was observed. This heating technique was used throughout this study.

Hydrogen brazing. - As mentioned earlier, the hydrogen brazing was done commercially. A dewpoint ranging from -70° to -80° F was reported. A complete set of shear specimens and sheet-metal specimens was brazed at 2150° F, which was held for 5 minutes. Although a dewpoint of -70° to -80° F was reported, the specimens were badly oxidized. This oxidation prevented the brazing alloy from wetting and flowing into the joints. Because of the resulting poor braze, these shear specimens were not tested. A group of shear specimens brazed at 2100° F and held for 30 minutes in a batch-type furnace was classed as satisfactory by the commercial operator and was tested. The sheet-metal specimens brazed at 2100° F and held for 30 minutes were also oxidized and were not tested.

A separate group of A-286 and Inconel X shear specimens was also brazed in a dry-hydrogen atmosphere. A technique used in hydrogen brazing of adding a flux to the brazing powder was used in place of plating the specimens.

Heat Treatment of Specimens

Since uniformity of processing and testing must be emphasized in a correlative study such as this, a single heat treatment was used for all specimens. An aging treatment of 1325° F for 16 hours was used for all shear and sheet-metal specimens after brazing. This single aging treatment, following a brazing cycle, produced near maximum strength in all four of the base metals.

Testing and Evaluation

All specimens were tested at 1200° F; the temperature was measured by a thermocouple in contact with the specimen. The specimens were held at temperature for 10 minutes prior to testing. A test temperature of 1200° F was selected since a range of 1000° to 1300° F is of primary interest to designers of air-cooled turbine blades. The method of testing shear specimens is shown in figure 6, while that for testing tensile specimens in a specially designed holder is shown in figure 7. The ends of the tensile specimens were preformed, as shown in figure 3, to fit the testing grips.

In order to compute the area of the shear joint, the length of engagement (see fig. 1) was measured for each shear specimen with depth and caliper micrometers before testing. All shear strengths (table III) reported were based on the entire area of engagement and not just on the actual area wetted and brazed. All specimens were checked visually after testing. The area for the tensile specimens was based on individual measurements for each specimen made prior to the application of brazing alloy. For the plated specimens, the area included the plate thickness.

RESULTS AND DISCUSSION

Vacuum Brazing

The results of testing over 350 specimens at 1200° F are presented in figure 8. The photomicrographs of figure 8 were made from sections of the fractured tensile sheet-metal specimens tested at 1200° F. All the results shown on this figure are from vacuum-brazed specimens.

The top row of photomicrographs in figure 8 shows the effect of brazing temperature and time at temperature on the base-metal structure with no braze applied. The tensile strength and elongation (percent over reduced section) are given, and the second row of photomicrographs shows the effect on these properties when the boron-bearing braze was applied. In addition, the shear strength values obtained from the shear specimens for the same brazing alloy and brazing conditions are given. The bottom row of photos in figure 8 shows the results when the boron-free brazing alloy was used.

As described previously, the braze was applied in a band 1/4 inch wide across the center of the 1-inch test section. During the braze heating cycle, the brazing alloys tended to flow along this 1-inch test section. When the specimens were tested in tension, all those coated with the boron-bearing braze fractured in the area where the braze had been initially applied. Those coated with the boron-free braze failed predominantly outside of the area where the braze had been applied initially (but still within the gage section). These data illustrate a general difference between the two braze alloys in regard to base-metal damage. All photomicrographs were taken of areas directly under that where braze had been initially applied and where penetration might be greatest. Thus photomicrographs for the boron-bearing braze show the fracture, whereas those photos for the boron-free braze generally do not.

It should be noted that the percent elongation was calculated on the basis of the entire reduced section. These values therefore do not represent, in all cases, the elongation that can be expected from the heavily brazed area. The tensile strength, elongation, and joint-shear-strength values are averages for the three specimens tested at 1200° F.

The photomicrographs and data missing from figure 8 resulted from specimens lost during brazing. The low melting electroless nickel plate brazed several specimens to the brazing fixture and the specimens could not be removed for evaluation.

Electrolytically plated A-286. - The results obtained from A-286 alloy electrolytically plated with nickel prior to brazing are shown in figure 8(a). A plate thickness of 0.0005 to 0.0008 inch was used on both the shear and sheet-metal specimens.

The photographs and data of A-286 with no braze reveal only a moderate effect of brazing temperature and time at temperature on the tensile strength and elongation of A-286. The tensile strength for the specimen processed at 2100° F for 15 minutes was 115,100 pounds per square inch, whereas it was reduced to 102,000 pounds per square inch after being held for 15 minutes at 2150° F.

The effect on the ultimate grain size of increasing time at high temperatures is pronounced. It can be seen in the top row of figure 8(a) that the grains are considerably larger for the longer holding times at brazing temperatures.

The photographs and data of A-286 with boron-bearing braze (fig. 8(a)) reveal the damaging effects of the boron-bearing braze. Both increased time and increased temperature have affected the structure as well as reduced the tensile properties. The reduction in tensile strength ranged from 6000 to 28,000 pounds per square inch. The brazing alloy nearly penetrated or diffused through the entire 0.030 inch of base metal when held at 2150° F for 15 minutes.

The data of A-286 with boron-free braze reveal that there was no appreciable effect on either the base-metal structure or the tensile properties.

Incomplete joint coverage by the boron-free braze was found when the lower brazing temperature and shorter time were used. For all the brazing cycles investigated, the boron-bearing braze produced shear joint strengths at least 15 percent greater than those for the boron-free brazing alloy.

Electroless nickel plated A-286. - In figure 8(b) the A-286 alloy was plated with 0.0003-inch electroless nickel prior to brazing. By comparing the results shown in the top row of figure 8(b) with those of figure 8(a), it can be seen that this plate itself can have a damaging effect on the structure and strength of the base alloy. This effect is most apparent for the longer times at brazing temperature. When compared with the same cycle for pure nickel plate, the decrease in tensile strength was as much as 28,000 pounds per square inch.

The boron-bearing braze caused a complete alteration of the base-metal structure during all brazing cycles. Although this resulted in considerable reduction in base-metal strength, the joint shear strengths for the two brazing alloys were nearly equivalent. These shear strengths, however, were found to average approximately 9000 pounds per square inch less than those for the electrolytic plated specimens (fig. 8(a)).

The data of boron-free braze with A-286 (fig. 8(b)) reveal not only that there is a less damaging effect than with the boron-bearing braze, but also that the boron-free braze prevents serious damage by the electroless nickel plate. Although it is not apparent from the photographs, the brazing alloy flowed over and wet the entire test section and prevented in some way the diffusion of the plate constituents.

Unplated Inconel X. - The results obtained from the unplated Inconel X specimens are shown in figure 8(c). Unplated Inconel X was included in this study because of the satisfactory flow and joint coverage found during preliminary tests. The preliminary tests were run in a 4-inch-diameter tube furnace prior to the completion of the induction furnace used in this program. None of the heating cycles used produced the complete joint coverage attained in the tube furnace. Although the pressures maintained in these furnaces are the same, the heating cycles are considerably different. Because of limited power available in the tube furnace, the heating through the melting range of the brazing alloy was very slow. The brazing alloy was estimated to be either partially or completely melted for at least 90 minutes. It appears that slow heating near the melting range aids the flow of the brazing alloy on Inconel X. It is possible that the increased time at high temperature aids the dissociation of the oxides on Inconel X.

Although both brazing alloys reduced the tensile strength somewhat, the boron-bearing braze has a greater detrimental effect on the elongation than the boron-free braze.

Electroless nickel-plated Inconel X. - In figure 8(d), the Inconel X was plated with 0.0003-inch electroless nickel prior to brazing. This phosphorous-bearing nickel plate, which melted at about 1700° F, affected the structure to varying depths for the brazing cycles used. In comparing the tensile strength of unplated and plated Inconel X, the top row in figures 8(c) and 8(d), respectively, a decrease of between 10 and 20 percent in tensile strength is attributable to the plating. As was the case with an A-286 alloy (fig. 8(b)), the specimens with the boron-bearing braze contained a seriously damaged area. Again, the boron-free braze resulted in negligible damage as compared with braze-free plated sheet, but produced shear joints much weaker than the boron-bearing braze.

Inconel X can also be plated by electrodeposition, which offers a more reliable method for vacuum brazing of this alloy. At the Lewis laboratory

at the present time unplated Inconel X is being vacuum brazed when small joints are involved, and electrolytically plated Inconel X vacuum brazed when large joints must be covered.

N-155. - The N-155 alloy (Multimet)(fig. 8(e)) requires no plating for either dry-hydrogen or vacuum brazing. No significant differences are apparent in the photographs and data of braze-free N-155.

The second row of figure 8(e) reveals a decrease of tensile strength from 78,000 to 60,000 pounds per square inch as time and temperature are increased. This decrease depicts the importance of close control of brazing temperature and time at brazing temperature when the boron-bearing braze is used.

As seen in the bottom row of figure 8(e), neither time nor temperature affect the strength of the sheet brazed with the boron-free braze. The boron-free braze affords a greater freedom of both temperature and time for the ranges investigated than the boron-bearing braze. The boron-bearing braze produced shear joints approximately 17 percent stronger than the boron-free braze.

L-605. - Considerable differences in strength and ductility were found with L-605 (AMS-5537) after the various brazing cycles (fig. 8(f)). The 2150° F brazing cycles caused more solutioning in L-605 than the 2100° F cycles. This difference is apparent in the top row of figure 8(f). All specimens were polished and etched in one mount. Although this technique may not have given an optimum etch for any one specimen, it revealed differences in metal structure effectively.

The boron-bearing braze decreased the base-metal strength and ductility; while the boron-free braze did not. Very close control of both temperature and time at temperature are necessary to minimize damage to base-metal structure and properties when thin sections are being brazed with the boron-bearing braze.

The shear joints on all the L-605 specimens were completely covered for both types of braze. The shear strengths for the boron-free braze were between 4 and 20 percent weaker than for the boron-bearing braze.

Hydrogen Brazing

The results of all the tested shear specimens brazed in dry-hydrogen for this study are shown in table III. All these specimens were brazed at the same holding time of 30 minutes at 2100° F in an atmosphere reported by the commercial operator to have a dewpoint of -70° to -80° F. A wide variation in joint coverage was found for the different specimens.

The results of all the tested shear specimens brazed in vacuum at 2100° F and held for 30 minutes were added to table III for convenient comparison. Whenever complete joint coverage was obtained, the shear strengths of the hydrogen processed specimens compared with completely covered joints produced during the same vacuum brazing cycle. The shear strengths of the L-605 specimens with both brazing alloys were almost identical to those brazed in vacuum. Again as was found in vacuum brazing, the boron-bearing braze was approximately 15 percent stronger than the boron-free braze.

Although every one of the unplated Inconel X and A-286 specimens that were brazed with flux contained visible flux inclusions, they were essentially as strong in joint shear strength as the best specimens produced in vacuum. In fact, this technique of brazing Inconel X was far superior to unplated or electroless nickel plate techniques evaluated in a vacuum. These same fluxes (calcium fluoride or sodium difluoride) volatilized in a vacuum atmosphere and were completely ineffective.

SUMMARY OF RESULTS

Vacuum and hydrogen brazing of four heat-resistant alloys with two types of high-temperature brazing alloy were investigated. Although it was thought that the primary difference between the brazing alloys was boron content, the boron-bearing alloy also contained a relatively high amount of carbon and some of the effects may be associated with this difference. The effect of time at two brazing temperatures (2100° and 2150° F) on the 1200° F shear strength of joints and on the base-metal tensile strength and elongation of four sheet materials (A-286, L-605, N-155, and Inconel X) was investigated.

The following results were obtained:

1. When boron-bearing braze was applied to the central area of the gage section of sheet tensile specimens, and the specimens were pulled in tension at 1200° F, as a result of braze penetration during the brazing cycles investigated, the tensile strength and ductility of the sheet were decreased appreciably. This base-metal damage became greater with increasing temperature and time of the brazing heating cycle. The amount of damage varied with the four base alloys.

2. When the boron-free braze was applied, the measured tensile strengths and ductilities of three base metals were generally the same as the specimens without braze but given the same respective braze heating cycle. In the case of Inconel X, some reduction in tensile properties was noted, but the reduction was less than for the boron-bearing braze.

3. Testing of shear joints at 1200° F of specimens brazed in vacuum and hydrogen revealed that joints brazed with the boron-bearing braze were somewhat stronger than joints brazed with boron-free braze. The difference varied for the four base metals brazed.

4. Shear specimens brazed in hydrogen with both types of brazing alloy exhibited more erratic joint coverage by the brazing alloy than those brazed in vacuum. The data indicated, however, if joint coverage was complete, that vacuum and hydrogen brazed shear specimens were equivalent in strength.

5. Alloys that can be age hardened and contain titanium and aluminum could be successfully brazed in vacuum by first electroplating with nickel. The results indicated that the addition of flux to the brazing alloy was more reliable for hydrogen brazing than was the plating. Flux additions were found to be ineffective during vacuum brazing.

6. The electroless type of nickel plate enhanced flow and joint coverage on alloys that could be age hardened and contained titanium and aluminum, but was found to have an adverse effect on base-metal strength and ductility as well as on joint shear strength.

Lewis Flight Propulsion Laboratory
National Advisory Committee for Aeronautics
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TABLE I. - CHEMICAL COMPOSITIONS OF BASE METALS AND BRAZING ALLOYS

	C	Mn	Si	Cr	Ni	Co	Mo	W	Cb	B	N ₂	Fe	Ti	Al	Cu	S	P	V
Brazing alloy																		
Boron-bearing braze	0.65		4.5	15	Balance					3.75		4.00						
Boron-free braze	0.14		10	19.5	Balance							1						
Base alloy																		
A-286 (AMS-5525)	0.08	1.5	0.7	14.5	25.5		1.2					Balance	2	0.3		0.03	0.04	0.3
Inconel X (AMS-5542)	0.08	0.7	0.5	14.5	70							7	2.5	0.7	0.2	0.01		
N-155 (AMS-5532b)	0.15	1	0.5	20	20	20	3.25	2.5	1		0.15	Balance						
I-605 (Haynes no. 25) (AMS-5537)	0.12	1.5	1	20	10	51		15				3						

TABLE II. - BRAZING VARIABLES INVESTIGATED

[Test temperature, 1200° F.]

Brazing alloy	Boron-bearing braze												Boron-free braze											
Plating alloy	Electroless nickel				Electrolytic nickel				No plate				Electroless nickel				Electrolytic nickel				No plate			
Brazing temp., °F	2100		2150		2100		2150		2100		2150		2100		2150		2100		2150		2100		2150	
Time at brazing temperature, min	15	30	5	15	15	30	5	15	15	30	5	15	15	30	5	15	15	30	5	15	15	30	5	15
Shear specimens ^a :																								
In vacuum:																								
A-286	x	x	x	x	x	x	x	x					x	x	x	x	x	x	x	x				
Inconel X	x	x	x	x					x	x	x	x	x	x	x	x					x	x	x	x
N-155									x	x	x	x									x	x	x	x
L-605									x	x	x	x									x	x	x	x
In hydrogen:																								
A-286		x				x								x				x						
Inconel X		x												x								x		
N-155										x												x		
L-605										x												x		
A-286 flux added										x												x		
Inconel X flux added										x												x		
Tensile specimens in vacuum ^b :																								
A-286	x	x	x	x	x	x	x	x					x	x	x	x	x	x	x	x				
Inconel X	x	x	x	x					x	x	x	x	x	x	x	x					x	x	x	x
N-155									x	x	x	x									x	x	x	x
L-605									x	x	x	x									x	x	x	x

^aThree shear specimens processed for each condition.^bSix tensile specimens processed for each condition, three with brazing alloy, three with no braze.

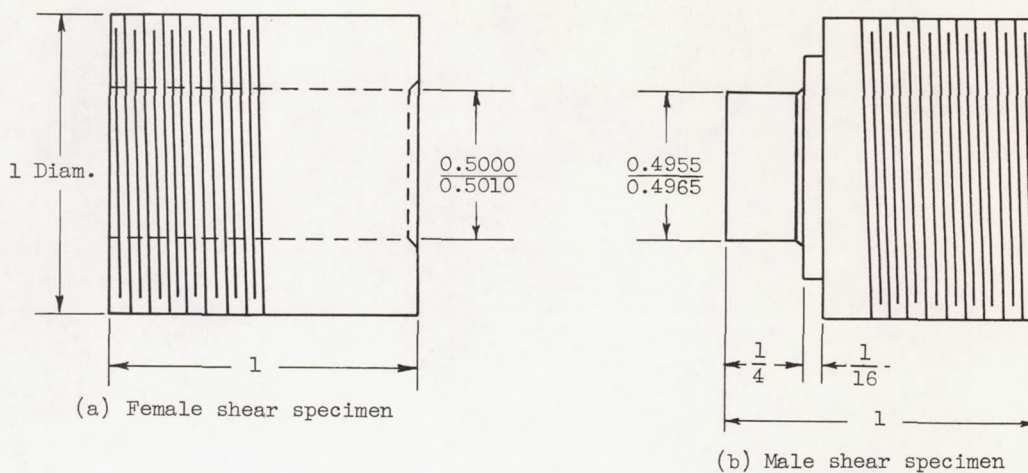
TABLE III. - COMPARISON OF 1200° F SHEAR JOINT STRENGTHS OF HYDROGEN-
AND VACUUM-BRAZED SPECIMENS

[Brazed at 2100° F for 30 min.]

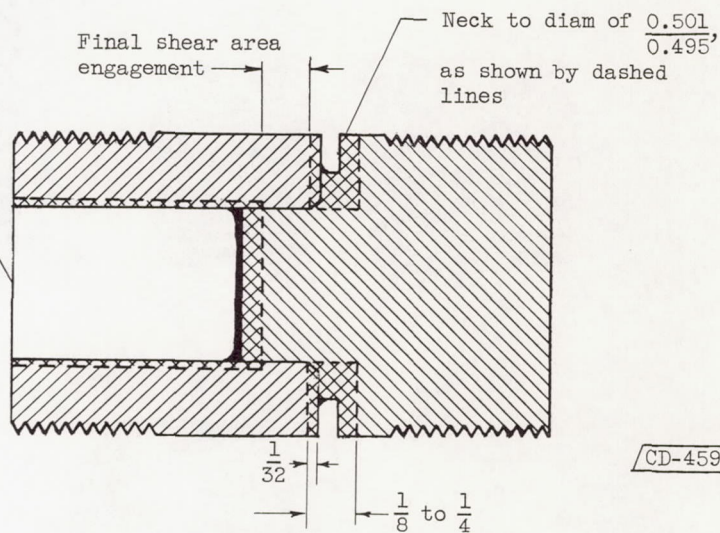
Alloy	Surface preparation	Shear strength, psi			
		Hydrogen		Vacuum	
		Boron-bearing braze	Boron-free braze	Boron-bearing braze	Boron-free braze
A-286	Electroless nickel plate	30,600	^a 15,300	31,600	33,600
		32,300	^a 10,500	31,900	32,600
		31,300	^a 20,100	32,000	32,900
A-286	Electrolytic nickel plate	^a 29,200	^a 25,300	38,900	34,200
		^a 26,700	^a 18,500	39,400	33,600
		^a 32,500	^a 20,200	38,700	34,200
A-286	Flux ^b added (unplated)	38,600	32,200	None	None
		37,500	31,000		
		40,200	30,500		
Inconel X	Electroless nickel plate	^a 18,500	^a 12,500	24,900	18,300
		^a 21,300	^a 10,000	23,100	18,500
		^a 16,500	^a 13,800	23,400	17,800
Inconel X	Flux ^b added (unplated)	35,800	29,300	^a 26,100	^a 27,700
		36,500	30,500	^a 25,700	^a 27,300
		35,200	31,500	^a 26,300	^a 28,100
				No flux	No flux
N-155	Unplated	^a 22,200	24,800	33,800	24,300
		^a 28,000	25,600	31,600	26,600
		^a 15,500	24,200	32,600	24,800
L-605	Unplated	38,600	30,200	38,800	32,900
		37,700	28,000	38,400	29,300
		35,600	31,800	38,100	31,300

^aIncomplete joint coverage; stress based on entire area of engagement.

^bAll flux treated specimens contained visible flux inclusions in the tested joints.



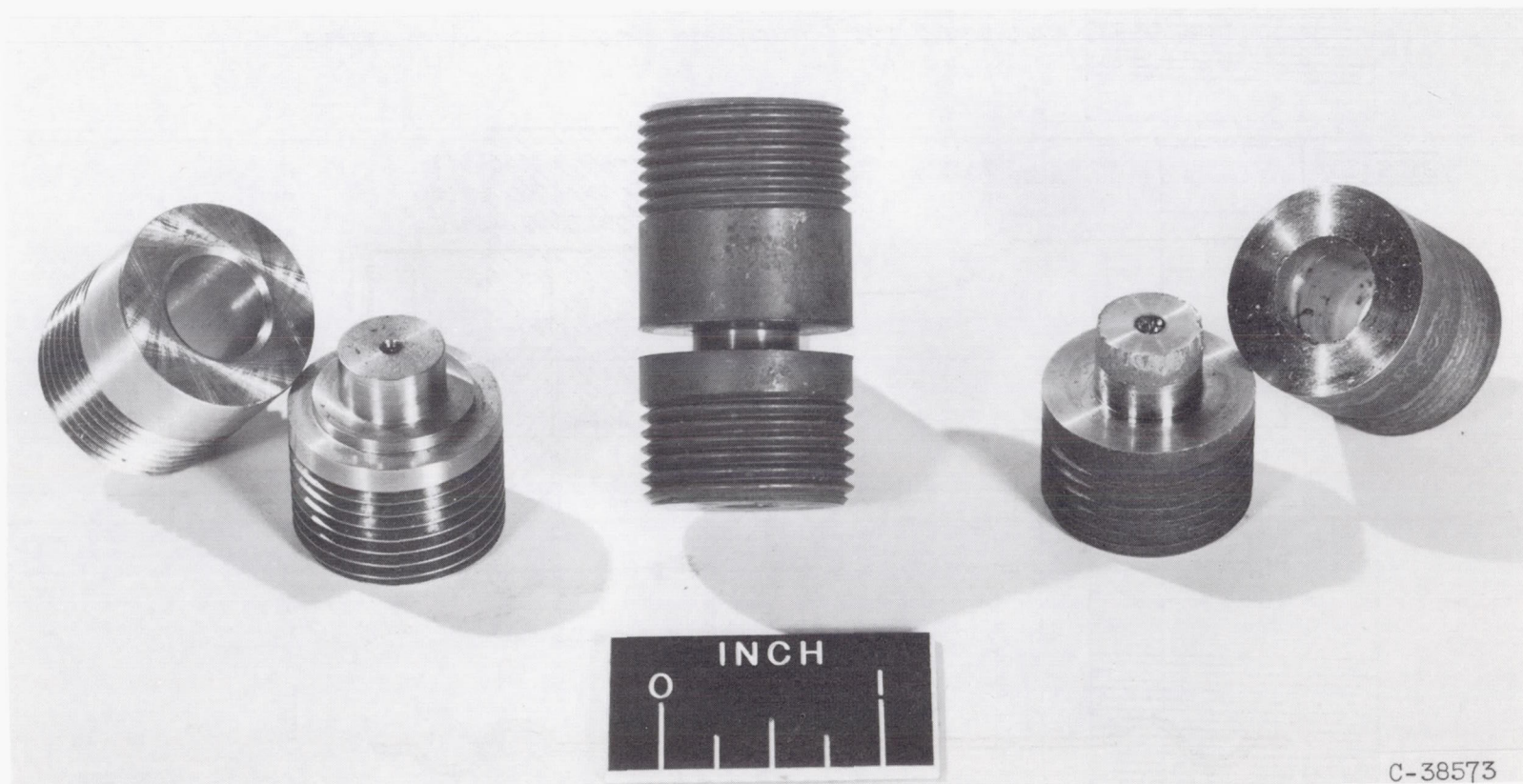
$\frac{17}{32}$ Flat bottom drill
to a depth of $\frac{51}{64} \pm 0.010$,
as shown by dashed lines



CD-4592

(c) Final machining of shear specimen after brazing

Figure 1. - Shear specimen. (All dimensions in inches.)



(a) Machined.

(b) Brazed and machined.

(c) Tested

Figure 2. - Shear specimens.

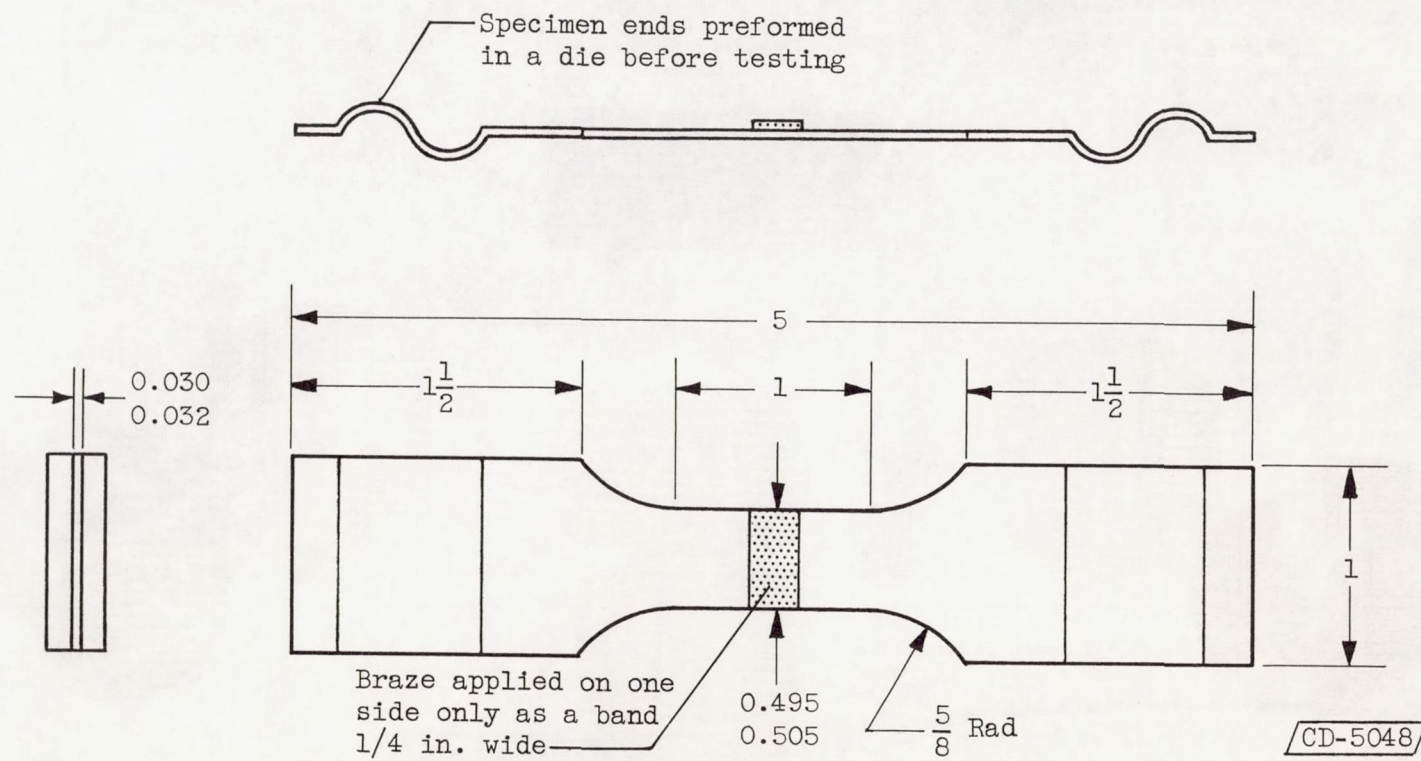


Figure 3. - Sheet-metal specimen. (All dimensions in inches.)

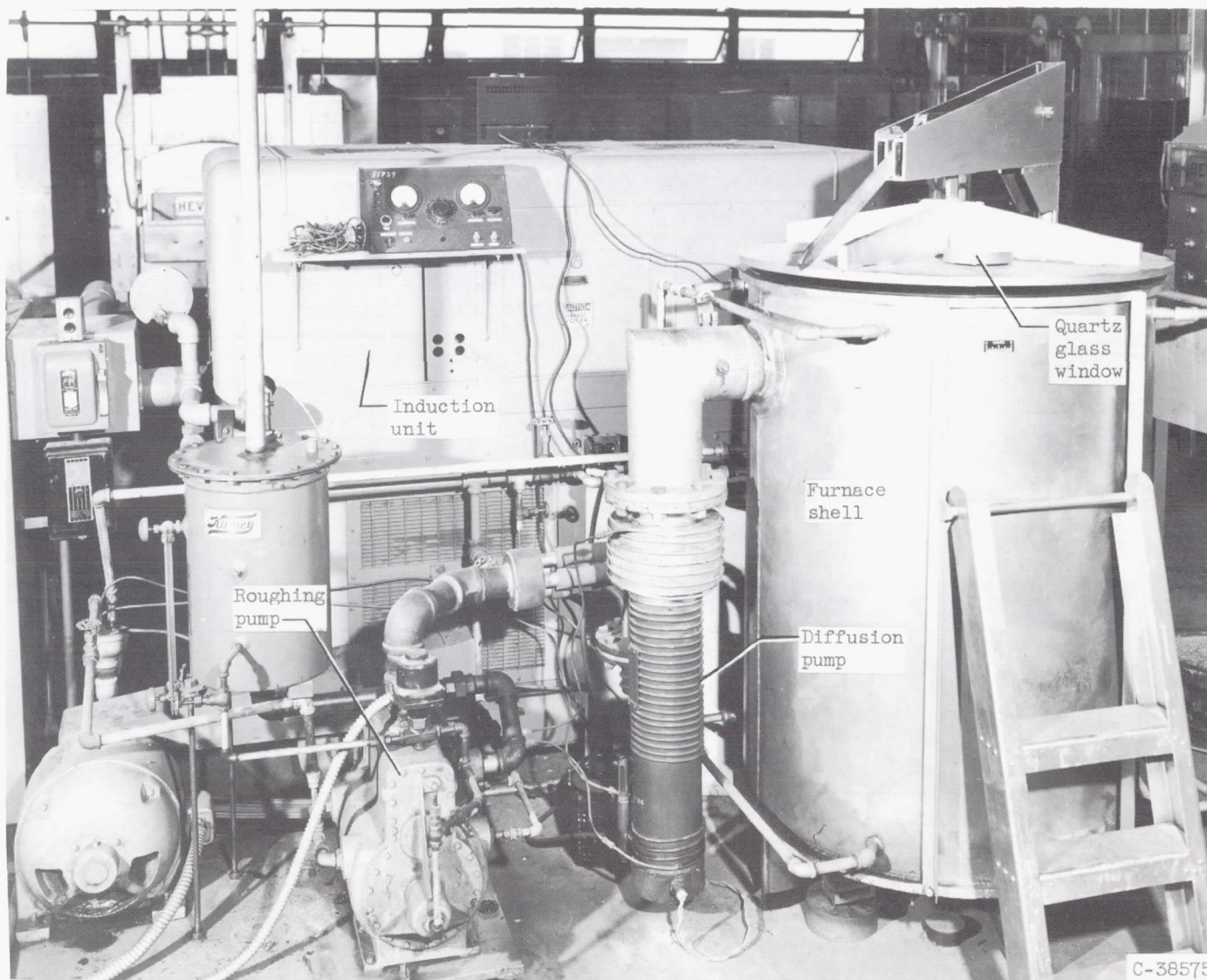


Figure 4. - Induction vacuum brazing furnace and equipment.

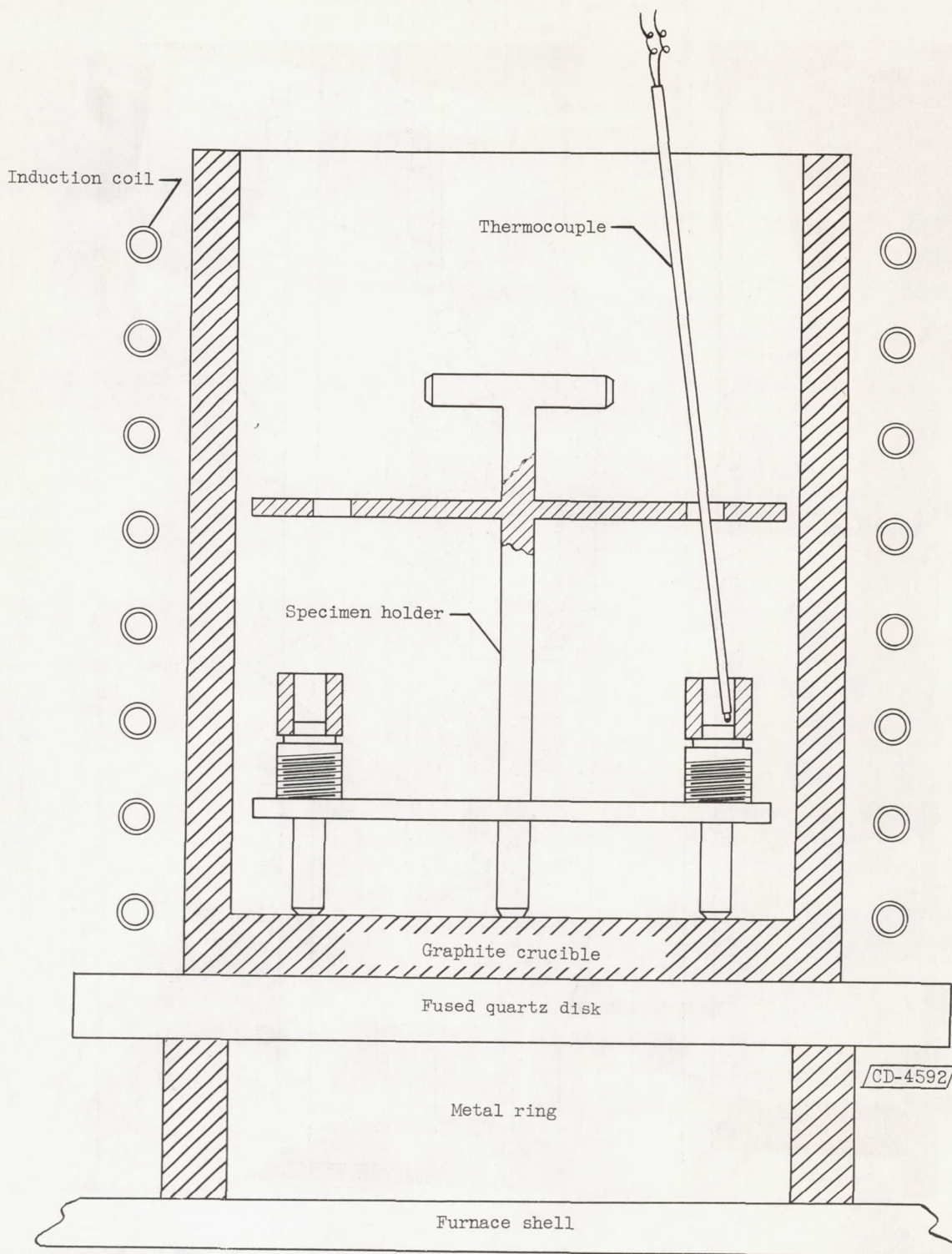


Figure 5. - Shear specimen placement within furnace crucible.

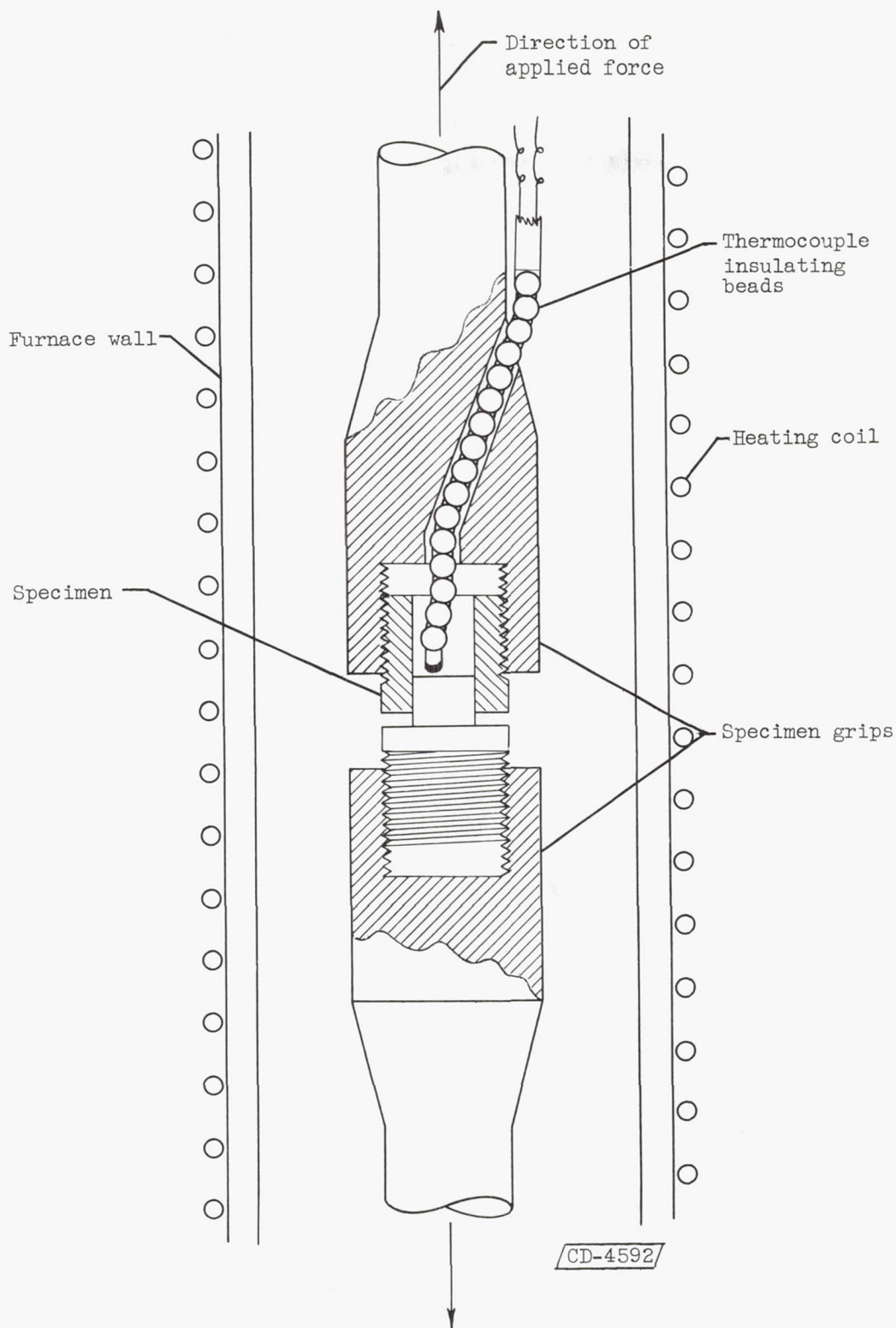
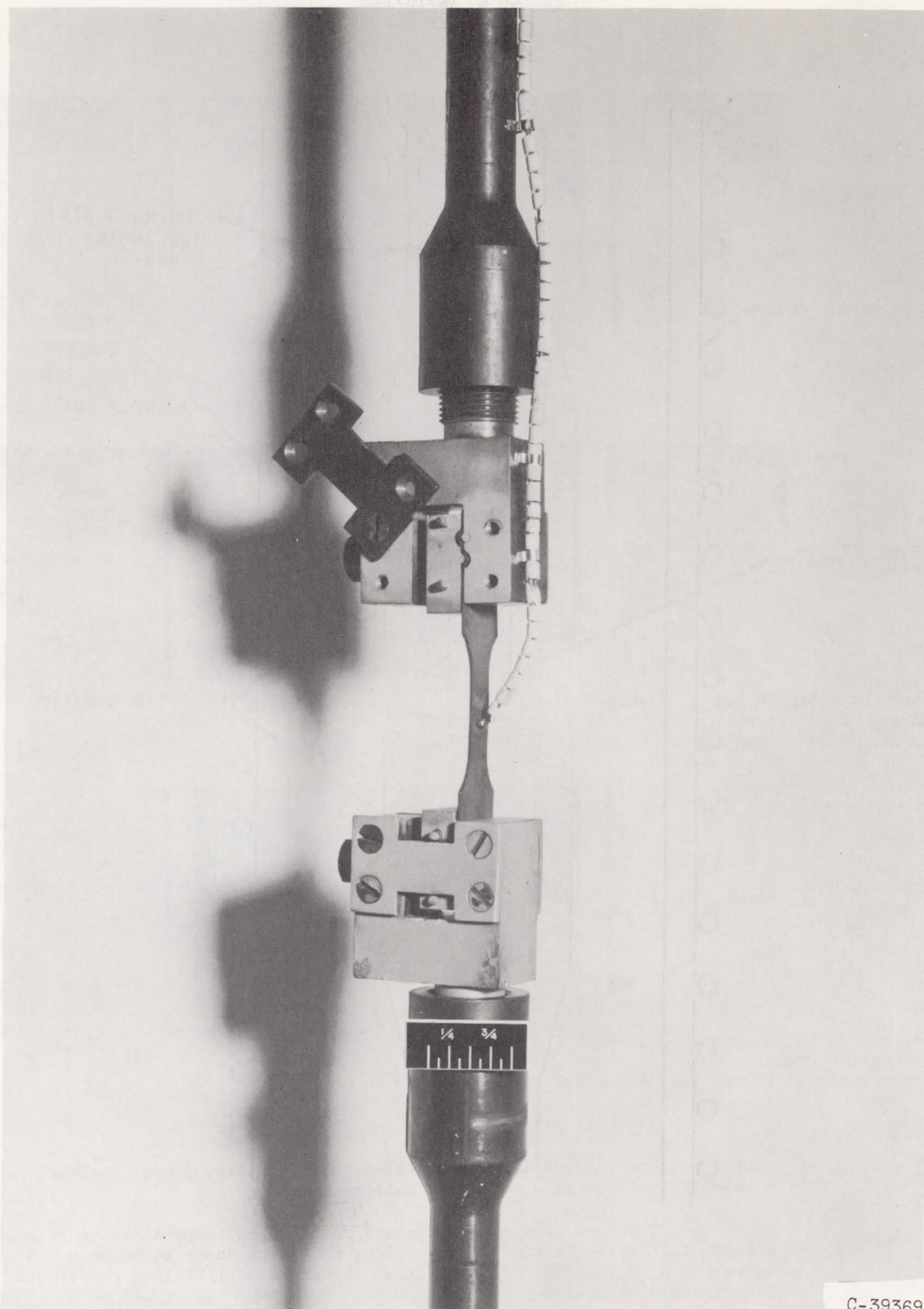


Figure 6. - Shear specimen setup for elevated-temperature testing.

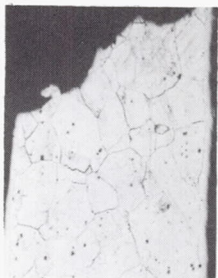


C-39369

Figure 7. - Gripping method of sheet-metal specimen in elevated-temperature testing setup.

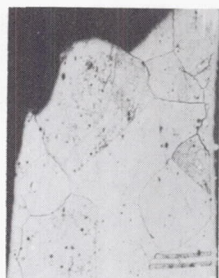
Brazing temperature, °F 2100
Time at temperature, min. 15

No braze



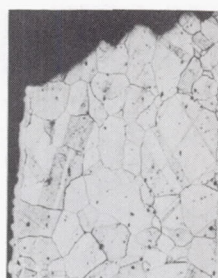
Tensile strength, psi 115,100
Elongation, percent 3

2100
30



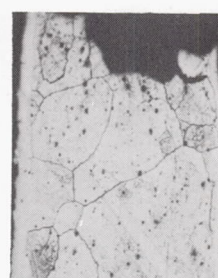
107,100
2

2150
5



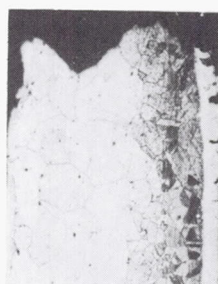
105,000
1

2150
15

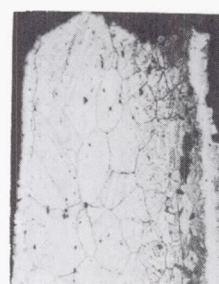


102,000
2

Boron-bearing
braze



Tensile strength, psi 99,000
Elongation, percent 0
Joint shear strength, psi 43,000



94,500
0
39,000



98,800
0
40,400

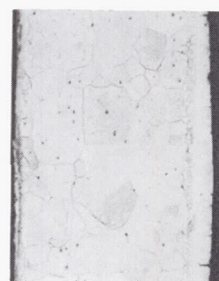


73,700
0
37,000

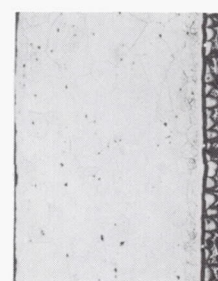
Boron-free
braze



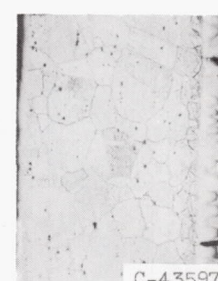
Tensile strength, psi 112,600
Elongation, percent 3^a
Joint shear strength, psi 20,000^a



106,500
2
34,000



115,100
3
33,000

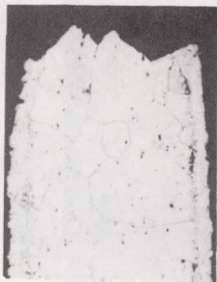
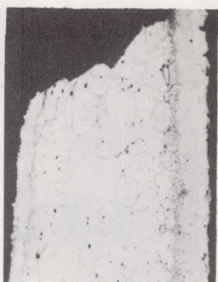
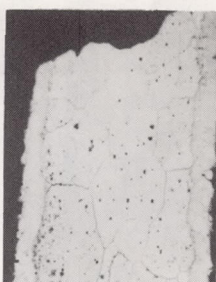
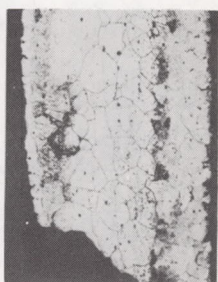
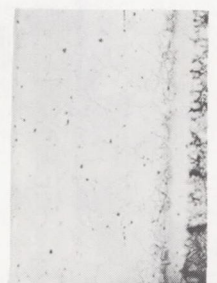


105,500
2
30,000

C-43597

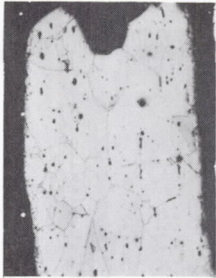
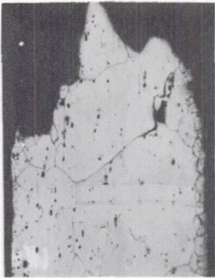
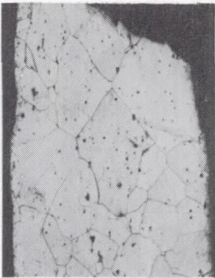
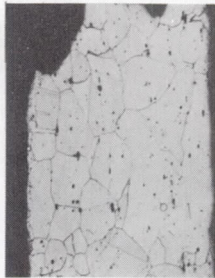
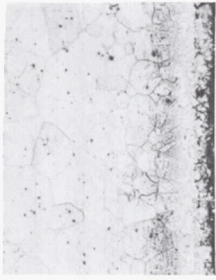
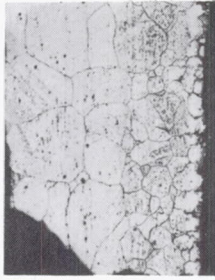
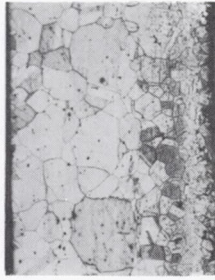
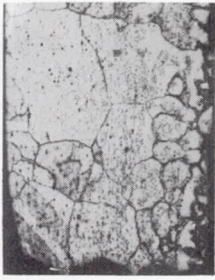
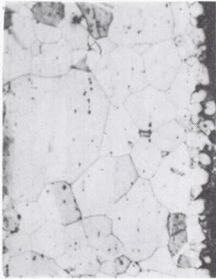
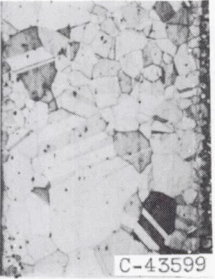
(a) Base metal, A-286 (AMS-5735); electrolytic nickel plate; etchant, electrolytic chromic acid plus Kallings reagent.

Figure 8. - Effect of brazing temperature and time at temperature on base-metal structure, tensile strength, elongation, and joint shear strength at test temperature of 1200° F. Full specimen thickness of 1/32 inch represented by photograph width; X100. ^aIncomplete joint coverage.

Brazing temperature, °F 2100		2100	2150	2150
Time at temperature, min. 15		30	5	15
No braze				
Tensile strength, psi	106,100	89,600	102,000	74,300
Elongation, percent	2	0	3	0
Boron-bearing braze				
Tensile strength, psi	87,000	79,500	72,000	55,000
Elongation, percent	0	0	0	0
Joint shear strength, psi	32,500	32,000	32,000	28,000
Boron-free braze				
Tensile strength, psi	110,000	102,000	112,000	95,000
Elongation, percent	3	3	5	4
Joint shear strength, psi	27,500	33,000	15,000 ^a	25,200

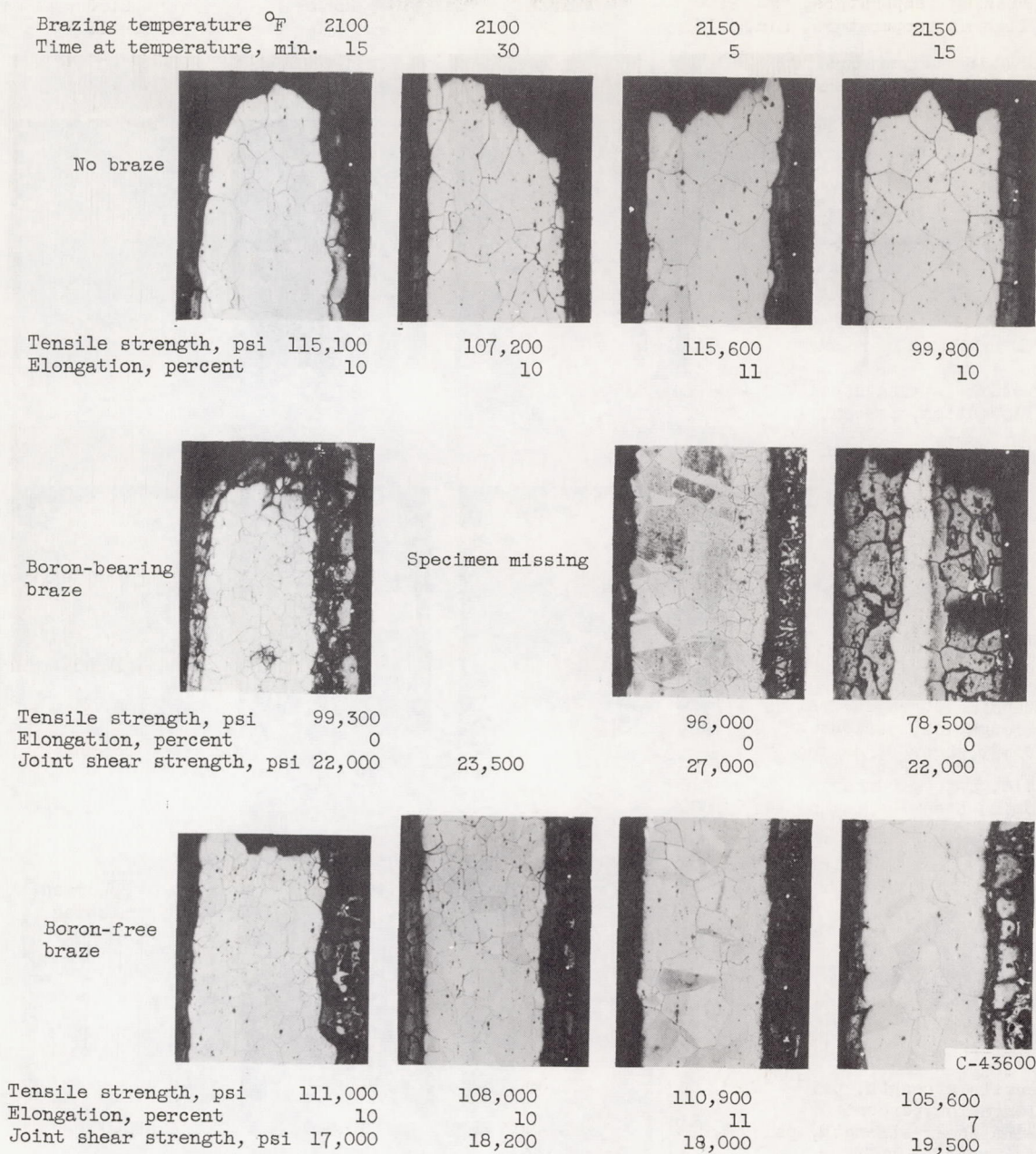
(b) Base metal, A-286(AMS-5735); electroless nickel plate; etchant, electrolytic chromic acid plus Kallings reagent.

Figure 8. - Continued. Effect of brazing temperature and time at temperature on base-metal structure, tensile strength, elongation, and joint shear strength at test temperature of 1200° F. Full specimen thickness of 1/32 inch represented by photograph width; X100. ^aIncomplete joint coverage.

Braze type		Brazing temperature, °F 2100 Time at temperature, min. 15				2100 30				2150 5				2150 15			
No braze																	
Tensile strength, psi		127,800		130,200		129,300		125,000									
Elongation, percent		12		12		10		10									
Boron-bearing braze																	
Tensile strength, psi		117,400		119,400		112,000		103,700									
Elongation, percent		2		3		0		0									
Joint shear strength, psi		15,000 ^a		26,100 ^a		24,500 ^a		23,800 ^a									
Boron-free braze								Specimen missing									
Tensile strength, psi		123,500		117,000		118,300											
Elongation, percent		15		13		12											
Joint shear strength, psi		9,200 ^a		27,600 ^a		10,100 ^a		16,000 ^a									

(c) Base metal, Inconel X (AMS-5542A); no plating; etchant, electrolytic chromic acid plus Kallings reagent.

Figure 8. - Continued. Effect of brazing temperature and time at temperature on base-metal structure, tensile strength, elongation, and joint shear strength at test temperature of 1200° F. Full specimen thickness of 1/32 inch represented by photograph width; X100. ^aIncomplete joint coverage.



(d) Base metal, Inconel X (AMS-5543A); electroless nickel plate; etchant, electrolytic chromic acid plus Kallings reagent.

Figure 8. - Continued. Effect of brazing temperature and time at temperature on base-metal structure, tensile strength, elongation, and joint shear strength at test temperature of 1200° F. Full specimen thickness of 1/32 inch represented by photograph width; X100.

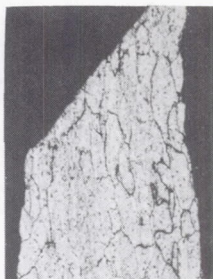
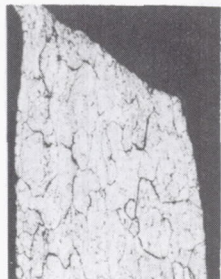
Brazing temperature, °F 2100
Time at temperature, min. 15

2100
30

2150
5

2150
15

No braze



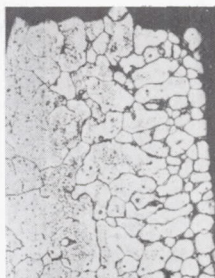
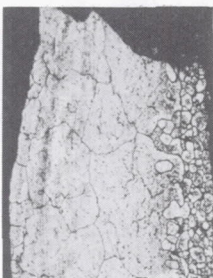
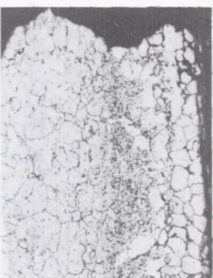
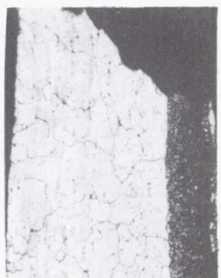
Tensile strength, psi 79,400
Elongation, percent 32

80,000
23

81,900
30

78,100
30

Boron-bearing
braze



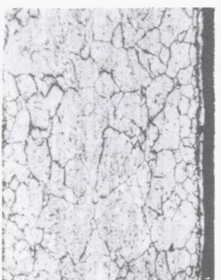
Tensile strength, psi 78,000
Elongation, percent 20
Joint shear strength, psi 33,000

72,000
12
32,500

69,000
12
32,000

60,000
10
32,000

Boron-free
braze



Tensile strength, psi 80,000
Elongation, percent 30
Joint shear strength, psi 30,000

81,500
25
25,500

81,000
30
27,000

82,000
30
28,500

C-43601

(e) Base metal, N-155 (AMS-5532b); no plating; etchant, Kallings reagent.

Figure 8. - Continued. Effect of brazing temperature and time at temperature on base-metal structure, tensile strength, elongation, and joint shear strength at test temperature of 1200° F. Full specimen thickness of 1/32 inch represented by photograph width; X100.

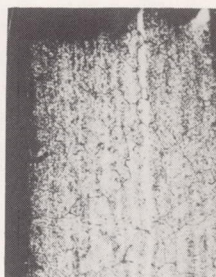
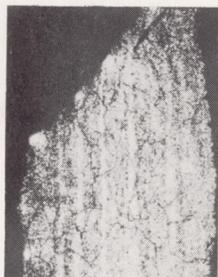
Brazing temperature, °F 2100
Time at temperature, min 15

2100
30

2150
5

2150
15

No braze



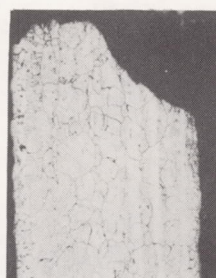
Tensile strength, psi 110,200
Elongation, percent 26

113,300
25

97,900
38

88,300
48

Boron-bearing
braze



Tensile strength, psi 80,000
Elongation, percent 19
Joint shear strength, psi 33,200

94,000
19
38,400

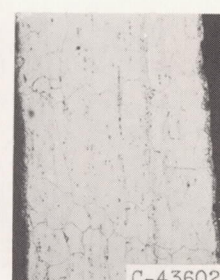
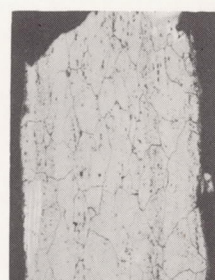
79,000
17
31,000

64,000
10
34,000

Boron-free
braze



SPECIMEN
MISSING



Tensile strength, psi 104,800
Elongation, percent 25
Joint shear strength, psi 32,500

31,200

96,000
38
29,500

87,000
38
27,300

C-43602

(f) Base metal, L-605 (Haynes 25); no plating; etchant, Kallings reagent.

Figure 8. - Concluded. Effect of brazing temperature and time at temperature on base-metal structure, tensile strength, elongation, and joint shear strength at test temperature of 1200° F. Full specimen thickness of 1/32 inch represented by photograph width; X100.